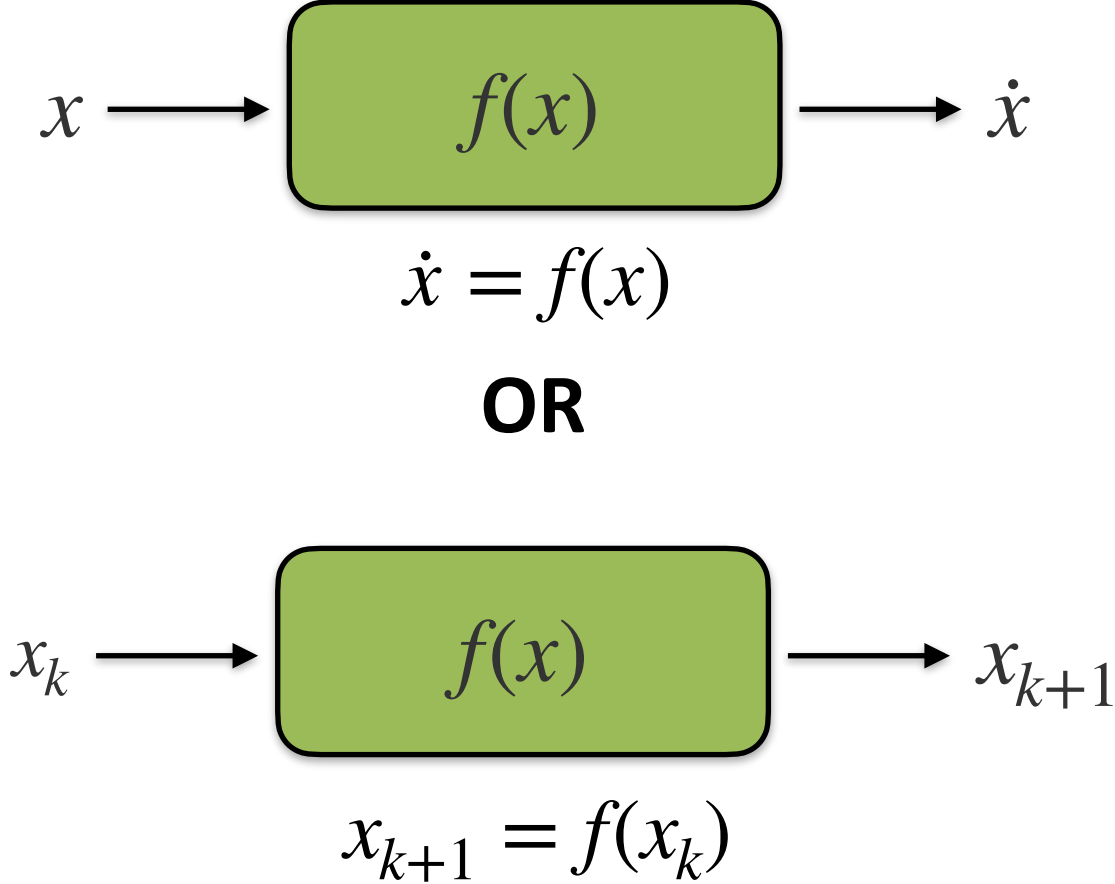


Actuator-State-Aware Koopman MPC

Zacharias Brown

Nonlinear Plant Model



A nonlinear $f(x)$ can be very unwieldy and thus make the future states hard to predict

A Different Approach

Instead of only watching x :

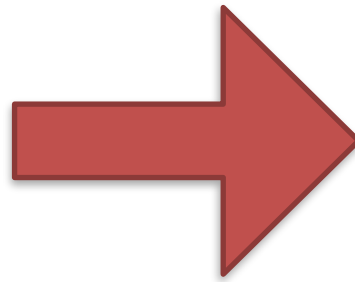
$$x = \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \\ \vdots \end{bmatrix}$$

$$x_{k+1} = f(x_k)$$

watch functions of x :

$$\begin{bmatrix} g_1(x) \\ g_2(x) \\ g_3(x) \\ \vdots \end{bmatrix}$$

$$g(x_{k+1}) = g(f(x_k))$$



Observables Become A Lifted State

Example Dictionary

$$z_k = \psi(x_k) = \begin{bmatrix} g_1(x_k) \\ g_2(x_k) \\ g_3(x_k) \\ \vdots \end{bmatrix}$$

$$\psi(x) = \begin{bmatrix} x \\ x^2 \\ \sin(x) \\ \cos(x) \\ \vdots \end{bmatrix}$$

The lifted state is a vector of chosen observables.

Polynomial Example

Take the plant

$$x_{k+1} = x_k^2$$

And lifted state

$$\psi(x) = \begin{bmatrix} x \\ x^2 \\ x^4 \end{bmatrix}$$

$$z_k = \begin{bmatrix} x_k \\ x_k^2 \\ x_k^4 \end{bmatrix} \quad \rightarrow \quad z_{k+1} = \begin{bmatrix} x_k^2 \\ x_k^4 \\ x_k^8 \end{bmatrix}$$

Two entries shift cleanly. One new entry appears.

Infinite Lift

With enough basis functions...

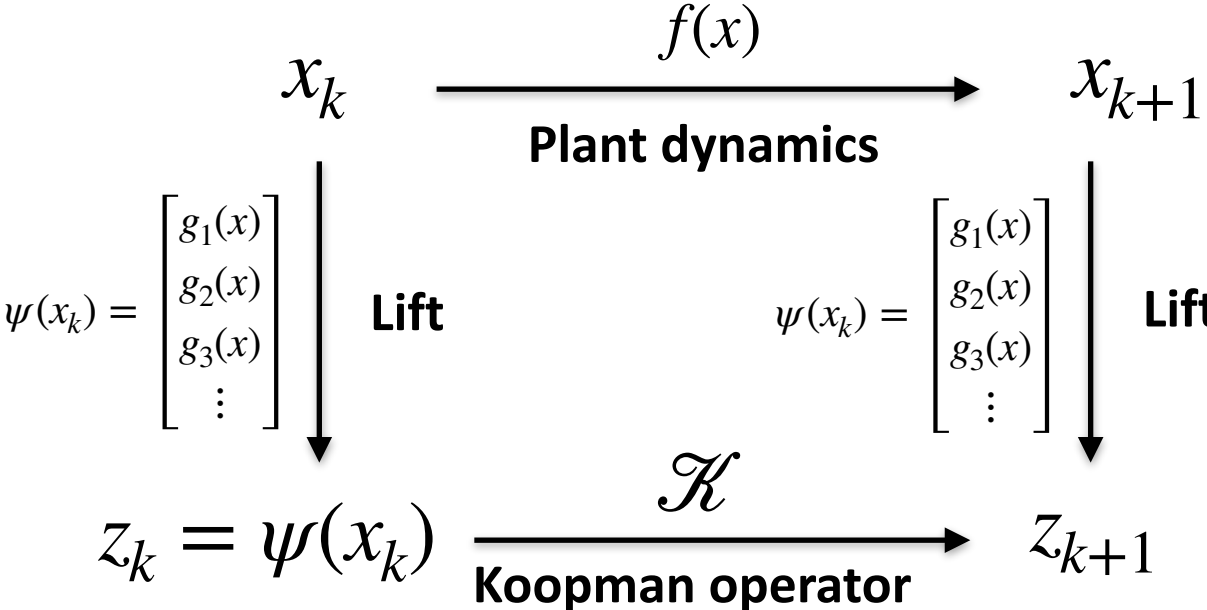
$$z_k = \begin{bmatrix} x_k \\ x_k^2 \\ x_k^4 \\ x_k^8 \\ x_k^{16} \\ \vdots \end{bmatrix} \quad z_{k+1} = \begin{bmatrix} x_k^2 \\ x_k^4 \\ x_k^8 \\ x_k^{16} \\ x_k^{32} \\ \vdots \end{bmatrix}$$

We approach a linear map from $z_k \rightarrow z_{k+1}$

$$z_{k+1} = \begin{bmatrix} 0 & 1 & 0 & 0 & 0 & \dots \\ 0 & 0 & 1 & 0 & 0 & \dots \\ 0 & 0 & 0 & 1 & 0 & \dots \\ 0 & 0 & 0 & 0 & 1 & \dots \\ 0 & 0 & 0 & 0 & 0 & \ddots \\ \vdots & \vdots & \vdots & \vdots & \vdots & \ddots \end{bmatrix} z_k$$

In fact, for a infinitely large lifted state, there is a liner mapping from $z_k \rightarrow z_{k+1}$

The Koopman Operator



$$(\mathcal{K}g)(x) = g(f(x))$$

The Koopman operator evolves the observables

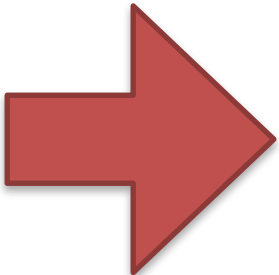
EDMD Finite Approximation

An infinitely large predictor isn't very helpful for control

$$z_k^{(\infty)} = \begin{bmatrix} g_1(x_k) \\ g_2(x_k) \\ g_3(x_k) \\ \vdots \end{bmatrix}$$

$$z_{k+1}^{(\infty)} = \mathcal{K} z_k^{(\infty)}$$

Infinite dimensional exact predictor.



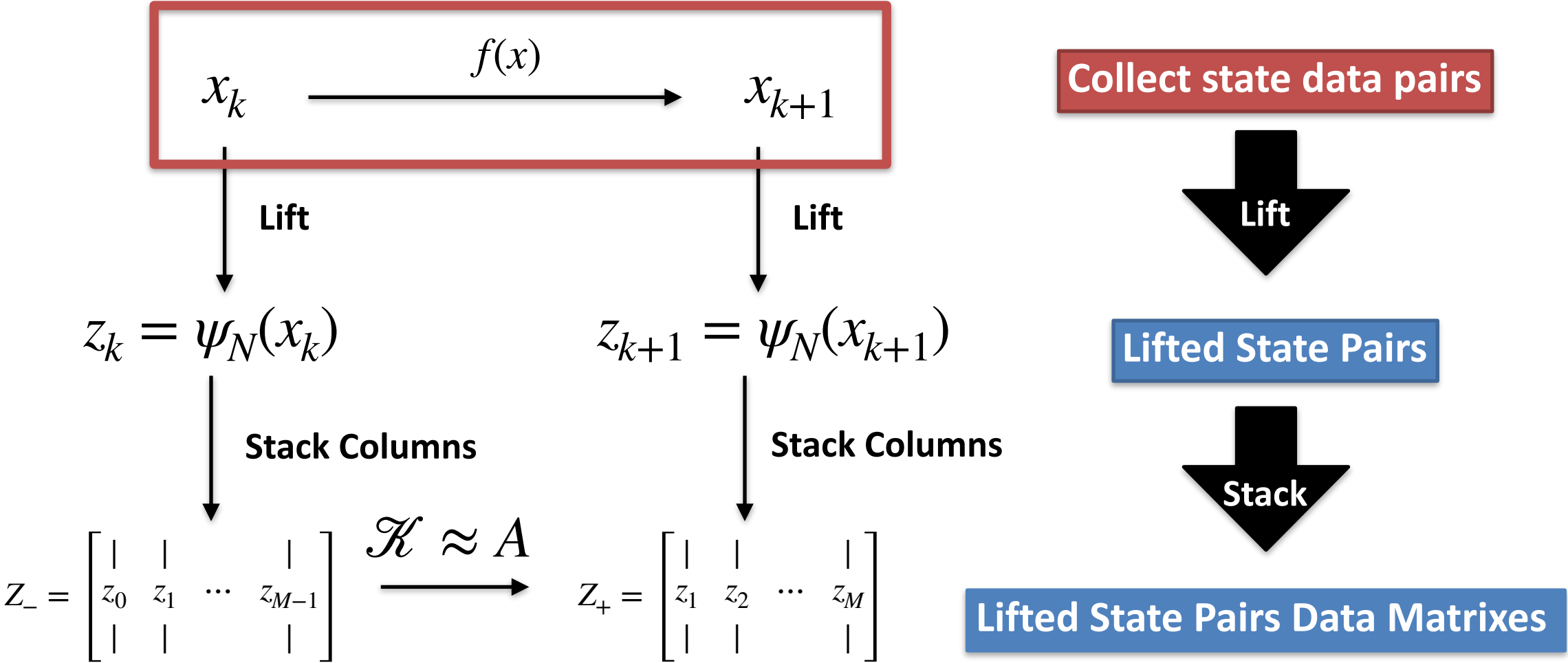
$$z_k = \psi_N(x_k) = \begin{bmatrix} g_1(x_k) \\ g_2(x_k) \\ \vdots \\ g_N(x_k) \end{bmatrix}$$

$$z_{k+1} \approx A z_k$$

$$\hat{x}_k = C z_k$$

Finite approximate predictor

EDMD Finite Approximation



Practical approximation: Extended Dynamic Mode Decomposition (EDMD).

EDMD linear regression fit

$$Z_+ \approx AZ_-$$

$$A^\star = \arg \min_A \|Z_+ - AZ_-\|_F^2$$

$$A^\star = Z_+ Z_-^\dagger \quad \text{or} \quad A^\star = Z_+ Z_-^T (Z_- Z_-^T + \lambda I)^{-1}$$

EDMD = finite lifted least-squares Koopman approximation.

Add Control: EDMDc / Koopman-MPC Form

$$x_{k+1} = f(x_k, u_k)$$

$$Z_+ \approx AZ_- + BU_-$$

$$z_{k+1} \approx Az_k + Bu_k$$

$$Z_+ \approx [A \quad B] \begin{bmatrix} Z_- \\ U_- \end{bmatrix}$$

$$U_- = \begin{bmatrix} | & | & & | \\ u_0 & u_1 & \cdots & u_{M-1} \\ | & | & & | \end{bmatrix}$$

$$[A^* \quad B^*] = \arg \min_{A,B} \left\| Z_+ - AZ_- - BU_- \right\|_F^2$$

$$[A^* \quad B^*] = Z_+ \begin{bmatrix} Z_- \\ U_- \end{bmatrix}^\dagger$$

Control term enters the lifted state **linearly**

The Input Channel Is Delicate

$$x_{k+1} = f(x_k)$$

↓

$$z_{k+1} = \mathcal{K} z_k$$

$$x_{k+1} = f(x_k, u_k)$$

↓

$$(\mathcal{K}_u g)(x) = g(f(x, u))$$

↓

$$z_{k+1} = \mathcal{K}_{u_k} z_k$$

↓

$$z_{k+1} \approx Az_k + Bu_k$$

Example

$$x_{k+1} = x_k^2 + x_k u_k$$

$$z_k = \begin{bmatrix} x_k \\ x_k^2 \\ x_k^4 \\ \vdots \end{bmatrix}$$

$$x_{k+1} = x_k^2 + \boxed{x_k u_k}$$

Nonlinear
Control
term

A constant B says the command has a fixed additive effect in lifted space.

That can be wrong when command input is not the force the plant actually feels.

Three Modeling Choices

1. Direct input

$$z_k = \psi(x_k)$$

$$z_{k+1} \approx Az_k + Bu_k$$

Simple, convex, but may hide actuator dynamics

2. Input features

$$\eta_k = \phi(u_k)$$

$$z_{k+1} \approx Az_k + B_\eta \eta_k$$

Richer input map, but more complex controller formulation

3. Add actuator state

$$x_{k+1} = f(x_k, a_k)$$

$$a_{k+1} = g(a_k, u_k)$$

$$z_k = \psi(x_k, a_k)$$

$$z_{k+1} \approx Az_k + Bu_k$$

Make the state more complete

Project choice: keep the command input simple, but expose the missing actuator state.

Actuator State Belongs In The Lift

Baseline Model

$$z_k^{\text{direct}} = \psi_x(x_k)$$

$$z_{k+1}^{\text{direct}} \approx A_d z_k^{\text{direct}} + B_d u_k$$

Actuator Aware Model

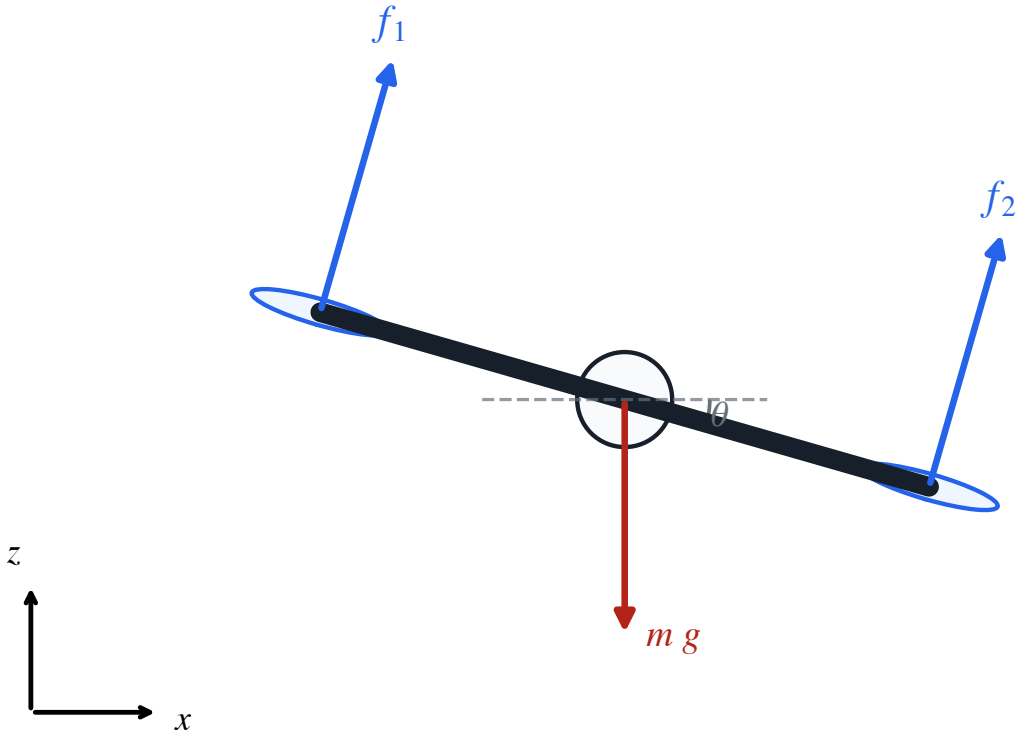
$$x_k^{\text{aug}} = \begin{bmatrix} x_k \\ a_k \end{bmatrix}$$

$$z_k^{\text{act}} = \psi_{xa}(x_k, a_k)$$

$$z_{k+1}^{\text{act}} \approx A_a z_k^{\text{act}} + B_a u_k$$

Does actuator-state lifting improve prediction and control under lag/saturation?

Plant Model



$$x = [p_x \quad \dot{p}_x \quad p_z \quad \dot{p}_z \quad \theta \quad \dot{\theta}]^T$$

Plant dynamics

$$\ddot{p}_x = \frac{\sin \theta}{m}(f_1 + f_2)$$

$$\ddot{p}_z = \frac{\cos \theta}{m}(f_1 + f_2) - g$$

$$\ddot{\theta} = \frac{d}{I_{yy}}(f_2 - f_1)$$

Actuator dynamics

$$\tau_m \dot{f}_1 = \text{sat}(u_1) - f_1$$

$$\tau_m \dot{f}_2 = \text{sat}(u_2) - f_2$$

$$u_i \longrightarrow \text{sat}(u_i) \longrightarrow f_i$$

Model Comparison

Direct Input

$$z_k^{\text{direct}} = \psi_x(x_k)$$

$$z_{k+1}^{\text{direct}} \approx A_d z_k^{\text{direct}} + B_d u_k$$

- Sees rigid body state only
- Commands u directly
- Does not know realized thrust f

Actuator-aware

$$z_k^{\text{act}} = \psi_{xf}(x_k, f_k)$$

$$z_{k+1}^{\text{act}} \approx A_a z_k^{\text{act}} + B_a u_k$$

- Sees rigid body state and realized thrust
- Commands u , but predicts F

Model Comparison: Function Basis

Direct Small

$$\psi_s(x_k) = \begin{bmatrix} p_{x,k} \\ \dot{p}_{x,k} \\ p_{z,k} \\ \dot{p}_{z,k} \\ \theta_k \\ \dot{\theta}_k \\ \sin(\theta_k) \\ \cos(\theta_k) \\ \dot{p}_{x,k}^2 \\ \dot{p}_{z,k}^2 \\ \dot{\theta}_k^2 \\ 1 \end{bmatrix} \in \mathbb{R}^{12}$$

Direct Matched

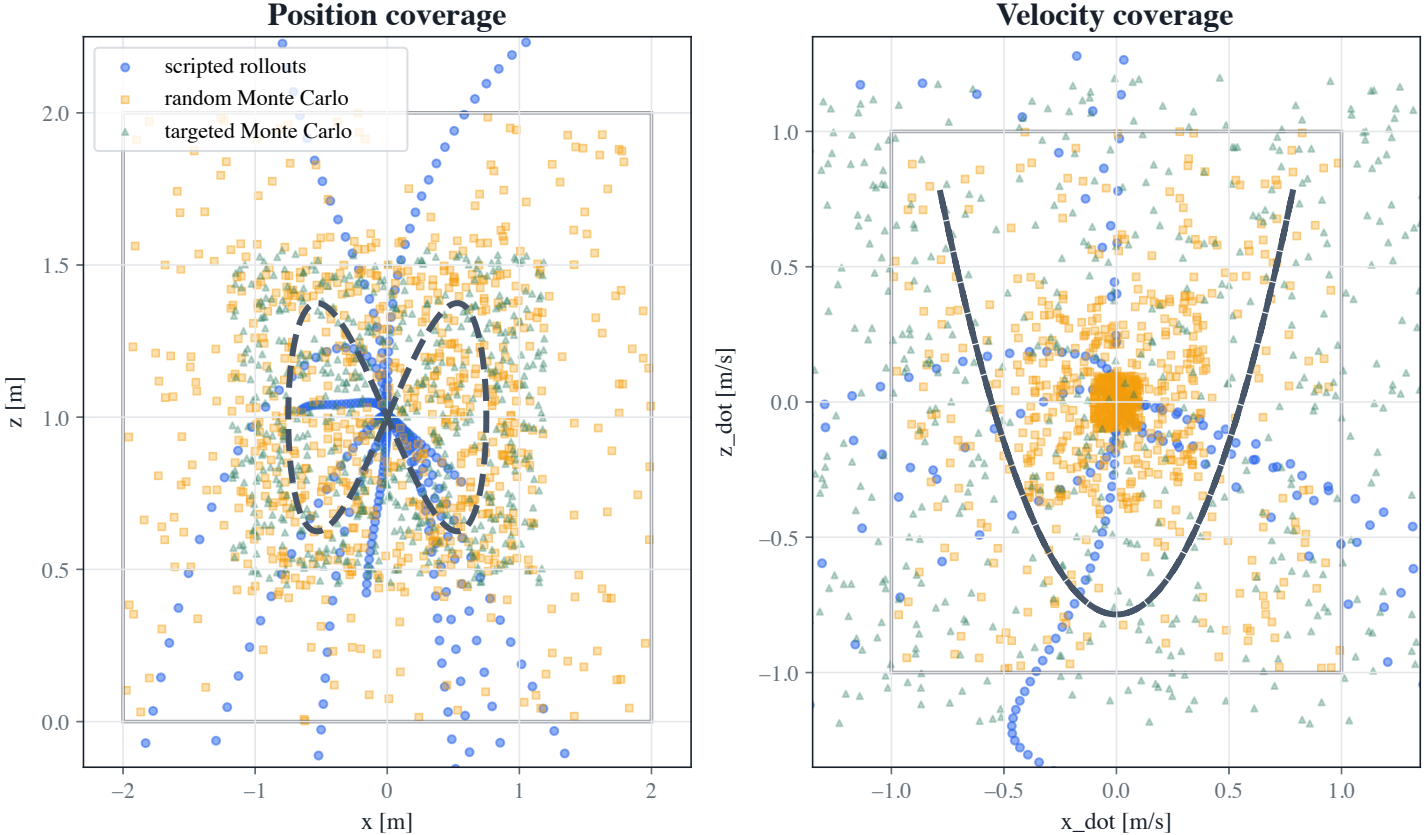
$$\psi_m(x_k) = \begin{bmatrix} p_{x,k} \\ \dot{p}_{x,k} \\ p_{z,k} \\ \dot{p}_{z,k} \\ \theta_k \\ \dot{\theta}_k \\ \sin(\theta_k) \\ \cos(\theta_k) \\ \dot{p}_{x,k}^2 \\ \dot{p}_{z,k}^2 \\ \dot{\theta}_k^2 \\ p_{x,k}^2 \\ p_{z,k}^2 \\ \theta_k^2 \\ p_{x,k}\theta_k \\ p_{z,k}\theta_k \\ \dot{p}_{x,k}\dot{\theta}_k \\ \dot{p}_{z,k}\dot{\theta}_k \\ 1 \end{bmatrix} \in \mathbb{R}^{19}$$

Actuator-aware

$$\psi_a(x_k, f_k) = \begin{bmatrix} p_{x,k} \\ \dot{p}_{x,k} \\ p_{z,k} \\ \dot{p}_{z,k} \\ \theta_k \\ \dot{\theta}_k \\ f_{1,k} \\ f_{2,k} \\ \sin(\theta_k) \\ \cos(\theta_k) \\ \dot{p}_{x,k}^2 \\ \dot{p}_{z,k}^2 \\ \dot{\theta}_k^2 \\ f_{1,k} + f_{2,k} \\ f_{2,k} - f_{1,k} \\ (f_{1,k} + f_{2,k})\sin(\theta_k) \\ (f_{1,k} + f_{2,k})\cos(\theta_k) \\ f_{1,k}f_{2,k} \\ 1 \end{bmatrix} \in \mathbb{R}^{19}$$

Data Collection

Training data coverage near the benchmark



“Random” Open-loop commands: hover steps, smooth sines, attitude pulses, near saturation.

Monte Carlo one-step samples cover the state/input box.

10,000 targeted figure-eight samples improve task-region accuracy.

MPC Controller

$$\min_{u_0, \dots, u_{N-1}} \sum_{j=0}^N \left\| Cz_j - r_j \right\|_Q^2 + \sum_{j=0}^{N-1} \left\| u_j - u_{\text{hover}} \right\|_R^2$$

With

$$z_{j+1} = Az_j + Bu_j$$

$$u_{\min} \leq u_j \leq u_{\max}$$

$$C_{\text{aug}} = \begin{bmatrix} I_8 & 0_{8 \times 11} \end{bmatrix} \quad z_k = \psi(x_k, f_k) =$$

$$Q = \text{diag}(20, 0.1, 20, 0.1, 0.01, 0.01)$$

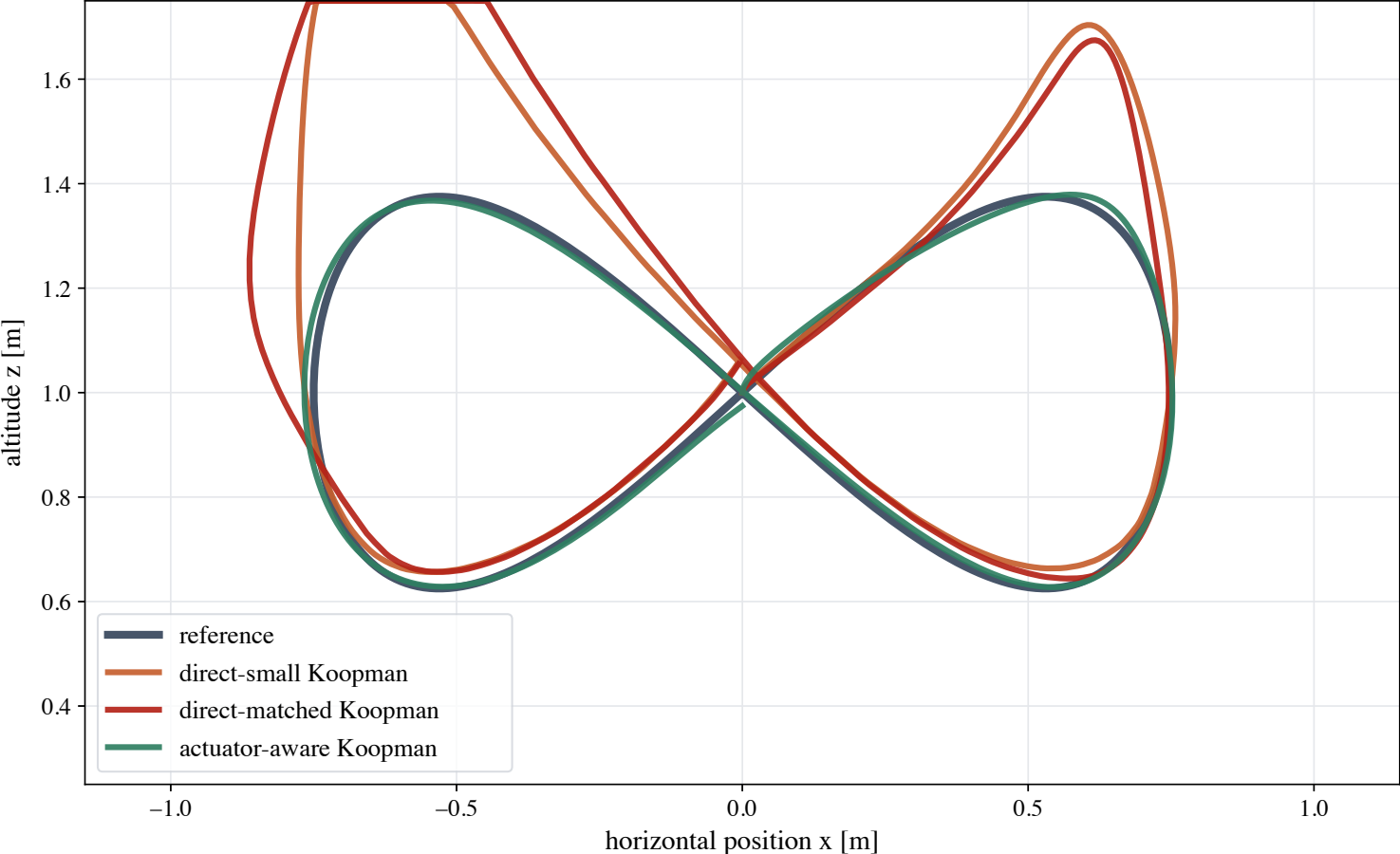
$$R = \text{diag}(0.03, 0.03)$$

$$N = 15$$

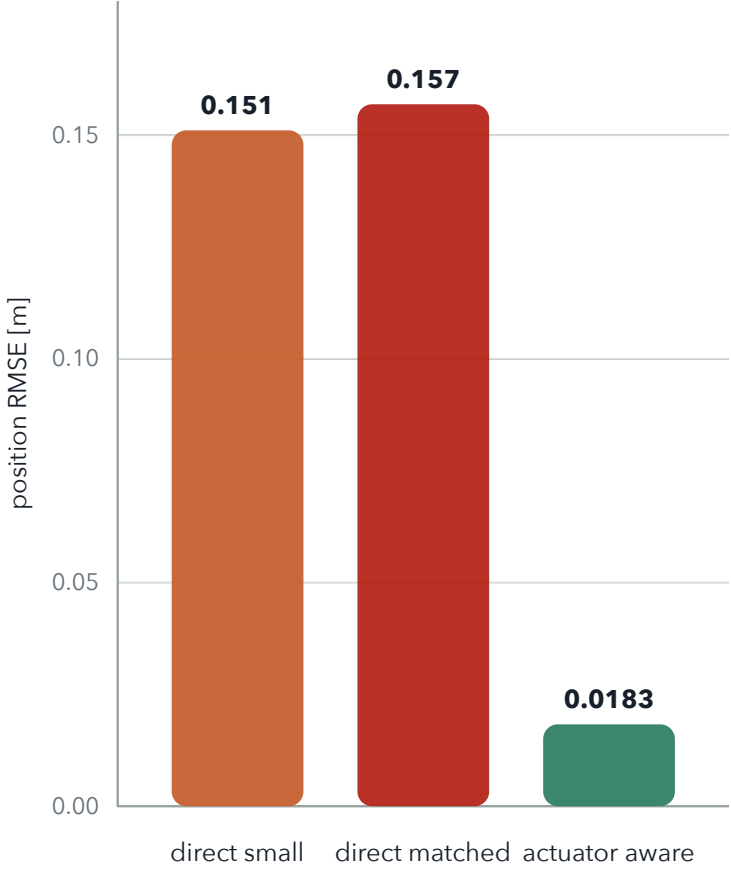
$$\begin{bmatrix} p_{x,k} \\ \dot{p}_{x,k} \\ p_{z,k} \\ \dot{p}_{z,k} \\ \theta_k \\ \dot{\theta}_k \\ f_{1,k} \\ f_{2,k} \\ \sin(\theta_k) \\ \cos(\theta_k) \\ \dot{p}_{x,k}^2 \\ \dot{p}_{z,k}^2 \\ \dot{\theta}_k^2 \\ f_{1,k} + f_{2,k} \\ f_{2,k} - f_{1,k} \\ (f_{1,k} + f_{2,k})\sin(\theta_k) \\ (f_{1,k} + f_{2,k})\cos(\theta_k) \\ f_{1,k}f_{2,k} \\ 1 \end{bmatrix} \in \mathbb{R}^{19}$$

Tracking Results

Figure 8 tracking with tau 0.02 seconds

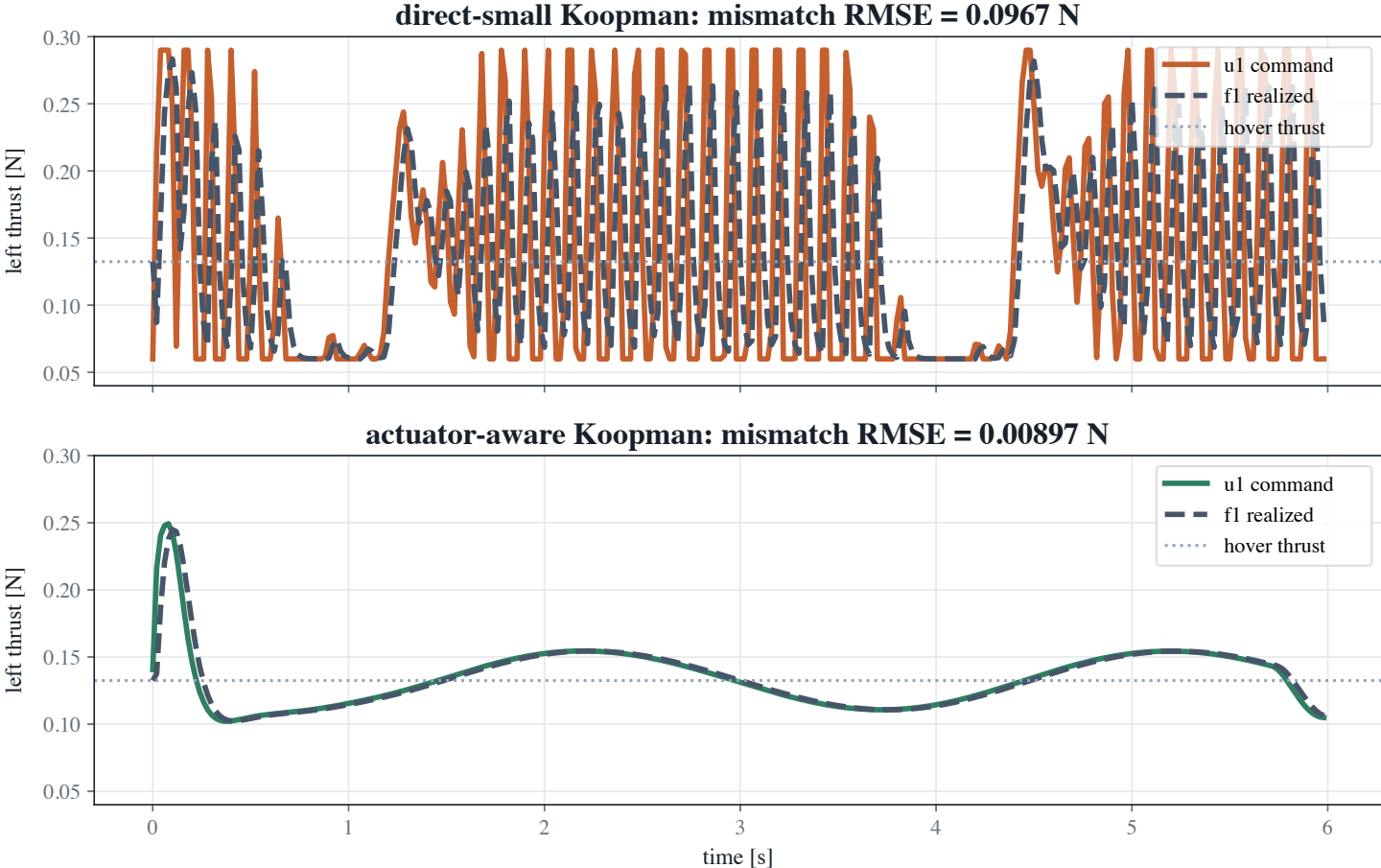


$$\tau_m = 0.02 \text{ s}$$



Actuator States

Commanded thrust versus realized thrust at $\tau_m = 0.02$ s



The actuator-aware model predicts the realized thrust state, so MPC can plan around the lag instead of fighting it

Lag Sensitivity

Tracking sensitivity to actuator time constant

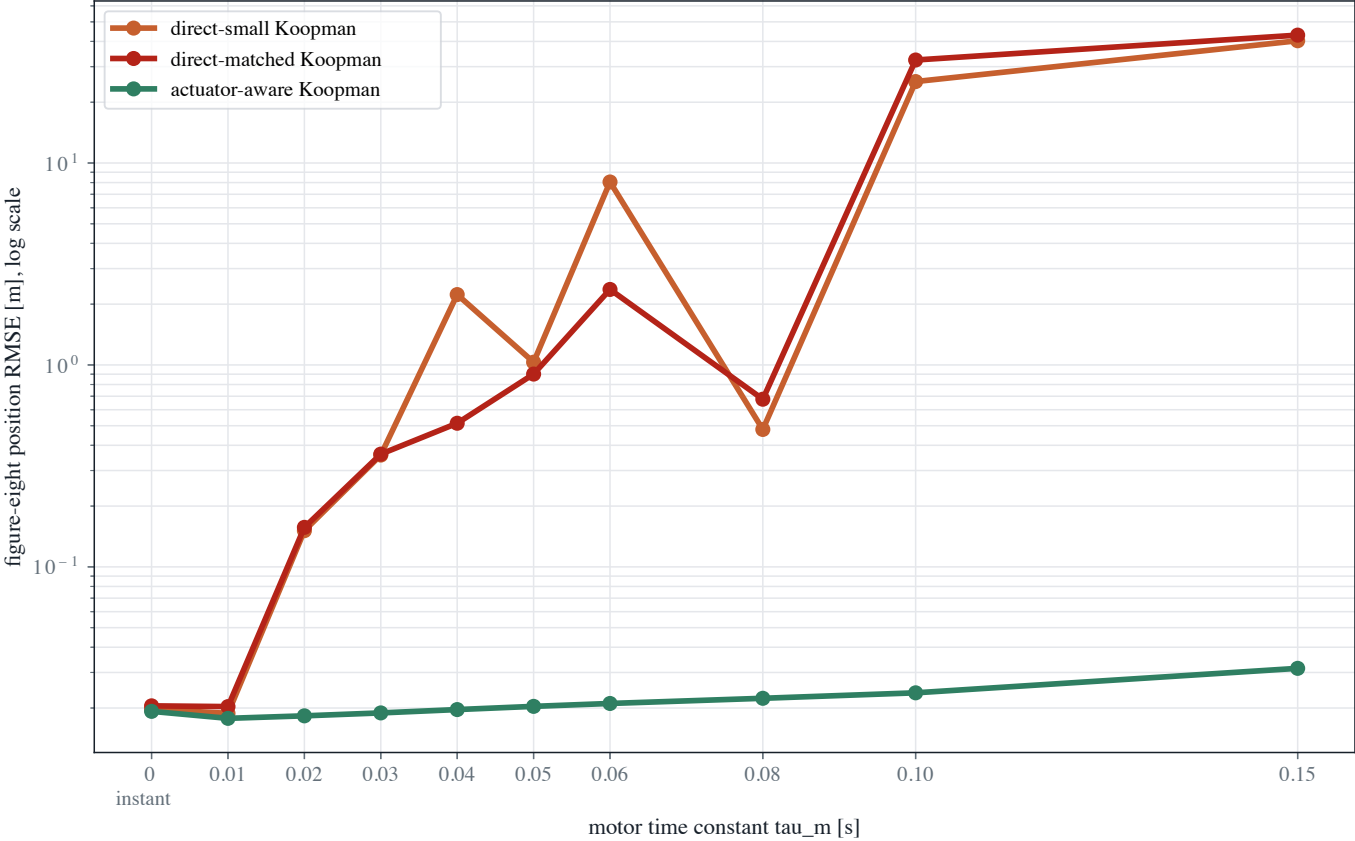


Figure-eight tracking with $\tau_m = 0.03$ s

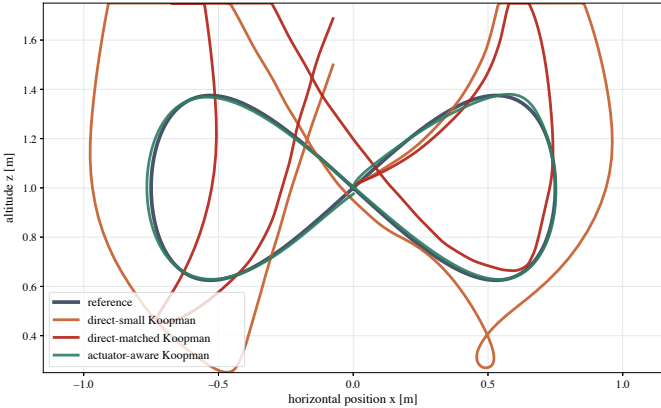
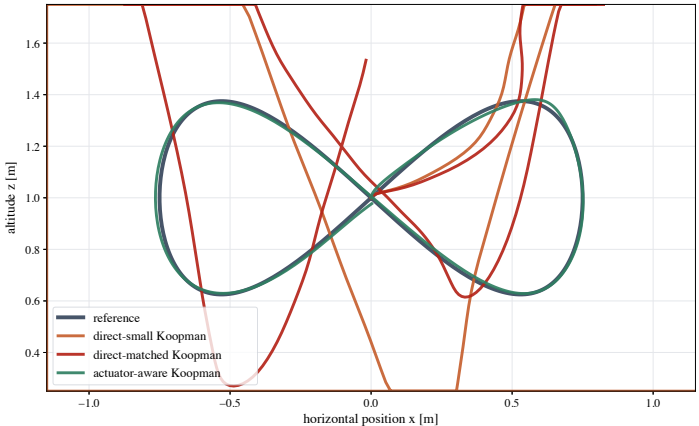


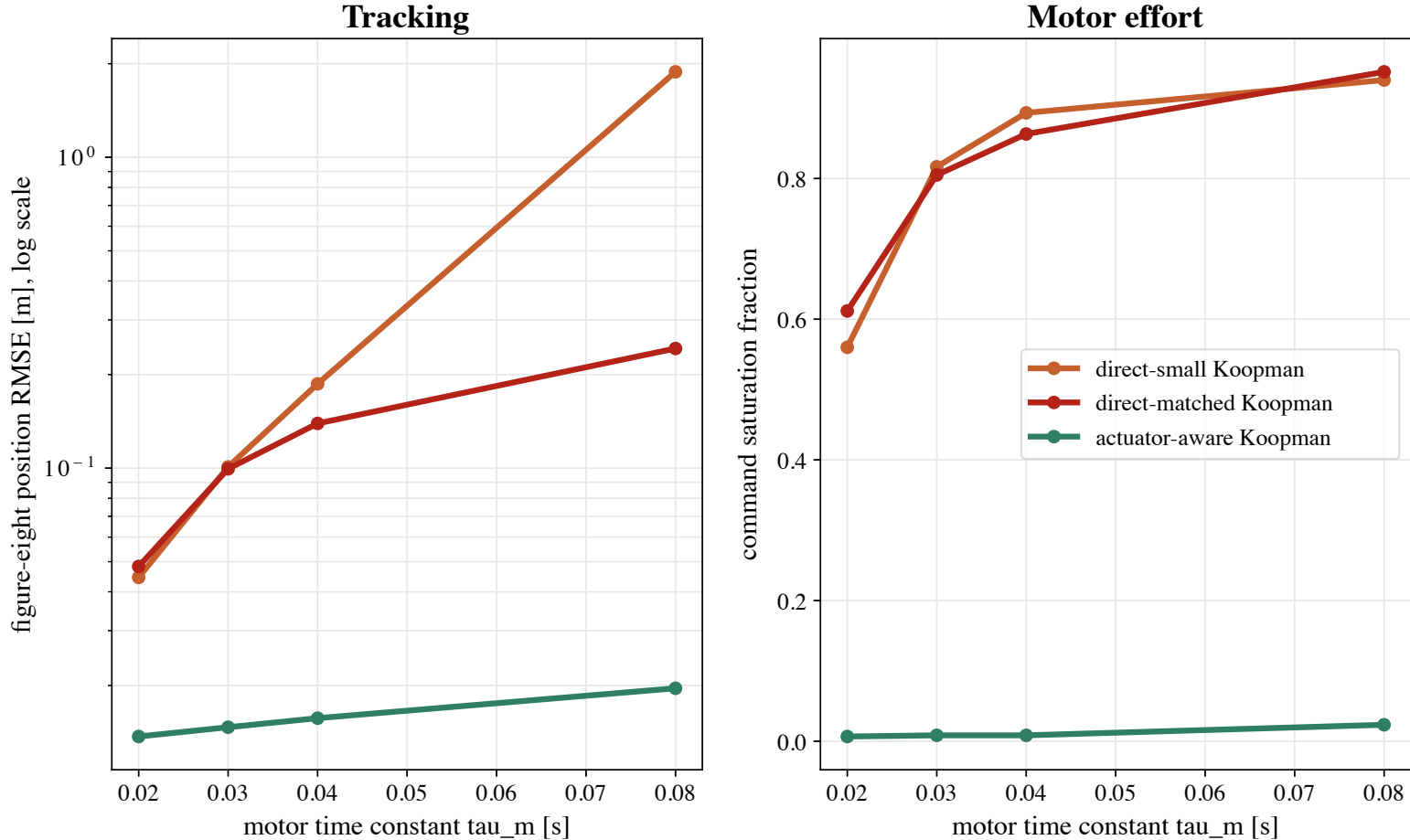
Figure-eight tracking with moderate actuator lag



When an actuator lag is very small, direct input approximation is less harmful, but as the actuator lag grows, modeling the actuator becomes exponentially more important.

Lag Sensitivity

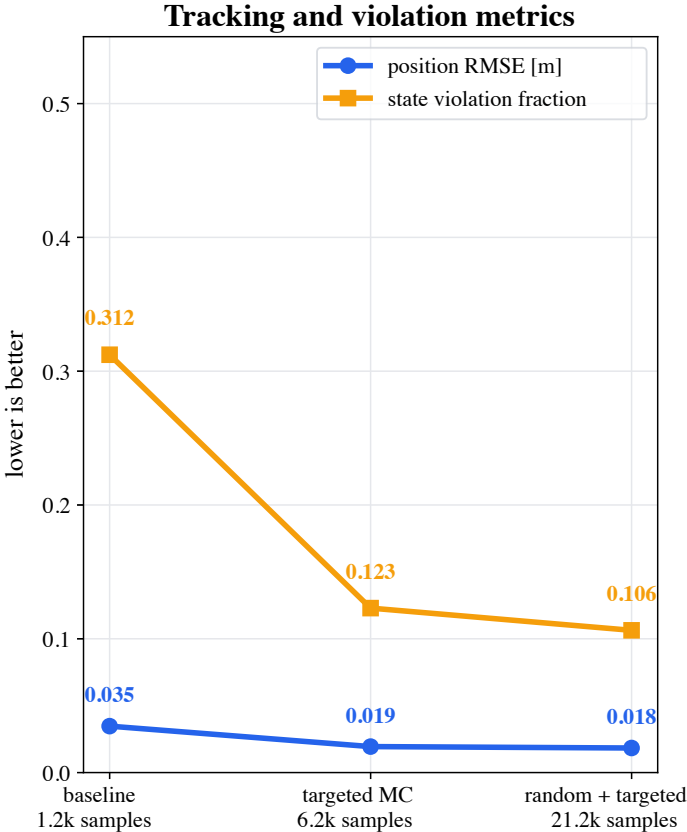
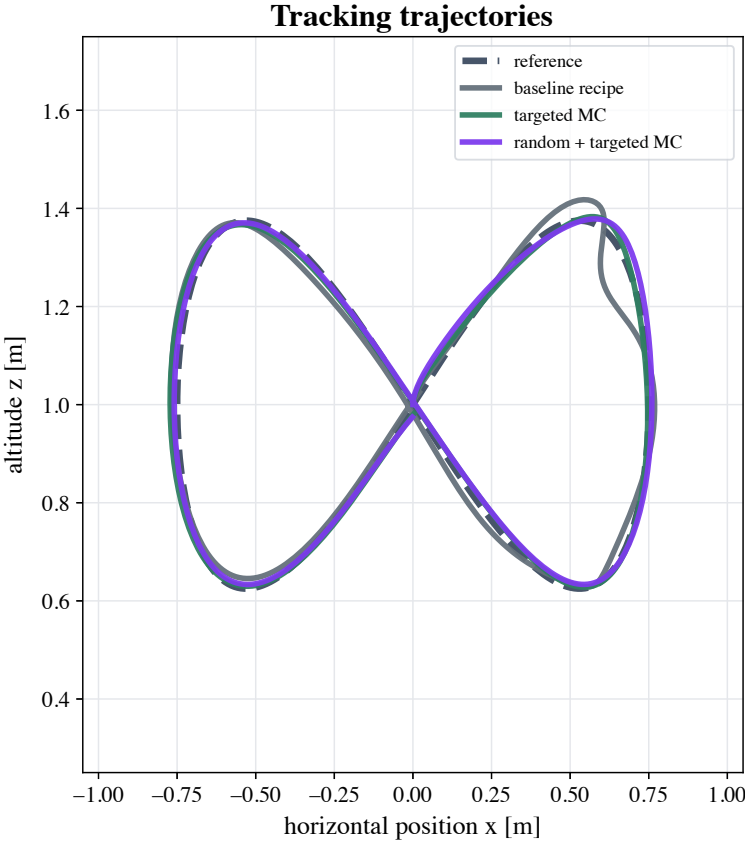
Best simple tuning across actuator time constants



Using tuning, we can make the larger basis non-actuator where it performs slightly better. More basis functions seems to help.

Data Matters

Data strategy comparison



The same Koopman-MPC architecture changes behavior depending on what data the least-squares fit sees.

- More data is better
- Data closer to your working region is better

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